

4 Apparatus for Testing Concrete under Multiaxial Compression 5 at Elevated Temperature (mac^{2T})

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11 **Abstract** This paper describes a new test facility for
12 determining material mechanical properties of struc-
13 tural concrete. The novel facility subjects 100 mm
14 cubic concrete specimens to true multiaxial compres-
15 sion ($\sigma_1 \neq \sigma_2 \neq \sigma_3$) up to 400 MPa at temperatures of
16 up to 300°C. Forces are delivered through three
17 independent loading frames equipped with servo-
18 controlled hydraulic actuators creating uniform dis-
19 placement boundary conditions via rigid platens.
20 Specimen deformation is calculated from displace-
21 ments measured to an accuracy of 10^{-6} m using a
22 system of six laser interferometers. The combination of
23 stiff loading frames, rigid platens, an accurate and
24 reliable strain measurement system and a fast control
25 system enables investigation of the material response
26 in the post-peak range. The in-house developed con-
27 trol software allows complex multi-stage experiments
28 involving (i) load and temperature cycling, (ii) small
29 stress probes and (iii) arbitrary (pre-defined) loading
30 paths. The program also enables experiments in which
31 the values of the control parameters and the execution
32 of the test sequences depend on the response of the
33 specimen during the test. The capabilities of the
34 facility are illustrated in this paper by experiments
35 determining the effects of different heat-load regimes
36 on the strength and stiffness of the material and tests
37 identifying the tangent stiffness matrix of the material
38 and the associated changes in the acoustic tensor under
39 multiaxial compression.

40 **Keywords** Test facility · Multiaxial compression ·
41 Concrete · Elevated temperature

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Introduction

The lack of a comprehensive database of stress and strain measurements on structural concrete under true multiaxial compression is one of the main obstacles preventing the development of reliable constitutive models for this important engineering material. Current design methods typically assume that the concrete in a column, beam or slab-like structural member is under essentially a uniaxial state of stress. The one-dimensional stress-strain relationship under such conditions is often idealised as following a parabolic form, broadly fitting data from uniaxial compression tests. However, in many safety-critical structures (such as nuclear reactor vessels, nuclear containment vessels, offshore platforms and arch-gravity dams) the stress state is most certainly not uniaxial throughout. Engineering precision and understanding is lost when the effects of confinement and the deformation behaviour in the other two principal directions are ignored. Furthermore, a constitutive model embracing the full three-dimensional behaviour is a requirement when attempting accurate continuum FE simulations.

Although the differences between the uniaxial and biaxial compressive strengths are not so marked, the fracture modes can be quite different [1-3]. Yet it is when all three principal stresses are compressive that very significant differences in strength, deformation and apparent ductility are seen. Most of the knowledge on the behaviour of concrete under multiaxial compression has been gathered from axisymmetric triaxial tests on cylindrical specimens. Triaxial rigs using the Hoek cell [4], originally developed for testing rock, provide simple and inexpensive devices that have proved popular for researchers investigating the response of concrete under confinement [5-7]. Newman's triaxial

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77 compression and extension tests [8], carried out in a
 78 specially designed apparatus operating at higher pres-
 79 sures, revealed the dramatic effect of confinement on
 80 the stress-strain nonlinearity of concrete. The strains he
 81 measured at peak stress were over 57 times larger than
 82 those seen under uniaxial compression. The triaxial cell
 83 tests performed by Jamet et al. [9] showed that with the
 84 increase of confinement the post-peak behaviour of
 85 micro-concrete gradually changed from brittle to duc-
 86 tile. Similar change in behaviour was observed in tests
 87 on concretes with three different strengths carried out
 88 in the Colorado triaxial cell [10]. These tests indicated
 89 that for higher strength concretes the transition from
 90 brittle to ductile behaviour occurred at higher levels of
 91 confinement. In 1999 Lee and Ansari [11] calibrated
 92 their constitutive model using data from tests on high-
 93 strength concrete in a cell allowing up to 83 MPa con-
 94 finement and 574 MPa axial stress. In 2002 Sfer et al.
 95 [12] tested larger (150 mm diameter, 300 mm long)
 96 cylinders in a triaxial cell with a capacity of 4.5 MN axial
 97 load and 140 MPa confinement pressure. These tests
 98 confirmed the transition from brittle to ductile behav-
 99 iour and suggested that with the increase of confine-
 100 ment the rupture mode changed from a diffuse
 101 distribution of microcracks to a mechanism involving
 102 fewer macrocracks separating the specimen into two or
 103 three blocks. This was in contrast to the earlier findings
 104 of Newman, which suggests that a particular care is ne-
 105 edded in interpreting the effects of the boundary con-
 106 ditions at the specimen-platen interface. Other triaxial
 107 cells introducing improvements of the original Hoek
 108 Cell design include the facilities in Milan and Bergamo
 109 [13] and the recently developed GIGA triaxial cell in
 110 Grenoble, a rig capable of delivering up to 2500 MPa
 111 axial stress on 70 mm diameter, 140 mm long cylindrical
 112 specimen, at confinements of up to 850 MPa [14].

113 While providing important information on the effect
 114 of confinement, the triaxial cells are constrained to
 115 operate on the compression and extension meridians
 116 (that is, the Lode angle is either $+\pi/6$ or $-\pi/6$). The
 117 first efforts in true multiaxial compression testing of
 118 concrete were made over thirty-five years ago by
 119 researchers at the University of New Mexico [15].
 120 They developed an apparatus comprising three inde-
 121 pendent frames with rigid platens, using polyethylene
 122 pads and grease to reduce the platen constraint.
 123 Manually operated, pressure controlled actuators with
 124 a load capacity of 270 kN were used to test 57 mm cubic
 125 specimens under stresses of up to 105 MPa. These
 126 findings were valuable, although comparisons with
 127 conventional triaxial results suggested that the New
 128 Mexico rig did not adequately remove platen friction. In
 129 the most comprehensive study of concrete under true

130 multiaxial compression, Scavuzzo [16] performed 67
 131 tests in a fluid platen rig at the University of Colorado.
 132 In addition to simple triaxial compression, triaxial
 133 extension and multiaxial compression with three differ-
 134 ent principal stresses, the cubic specimens were tested
 135 under cyclic, staircase, piecewise-uniaxial and circular
 136 loading paths. However, the Colorado stress-controlled
 137 rig was unable to capture any possible strain softening
 138 in the material. The stress range of this impressive
 139 apparatus was further limited by the use of leather pads
 140 inserted between the fluid cushions and the concrete
 141 specimens, which led to stress concentration and devel-
 142 opment of diagonal cracks near the corners of the
 143 specimens at relatively low stress levels. It was at the
 144 University of Eindhoven that the post-peak response of
 145 conventional structural concrete was captured using
 146 brush platens within a rig employing three suspended
 147 independent loading frames [17]. Although only a few
 148 experiments close to the biaxial regime were reported,
 149 that careful work clarified the role of local fragmenta-
 150 tion and macroscopic dilation on the loss of load-
 151 carrying capacity. The results obtained from using
 152 different boundary conditions in the Eindhoven rig
 153 suggested that PTFE-coated platens provided the
 154 lowest friction restraint in the post-peak region [18].

155 If there exists limited multiaxial data under ambient
 156 conditions, then the position is far worse under
 157 elevated temperature. The majority of the experiments
 158 performed on hot specimens have been restricted to
 159 uniaxial compression. The exceptions are the biaxial
 160 studies performed in Braunschweig [19, 20] and the
 161 8 torsional confined cylinder tests at Northwestern
 162 University [21]. In the latter tests, the aim was to
 163 identify the short-term creep response under controlled
 164 moisture conditions at temperatures of up to 200°C.
 165 Cylindrical specimens with 152 mm diameter were
 166 loaded under axial compression (< 10 MPa), lateral
 167 confinement (< 10 MPa) and torsionally induced shear
 168 (< 4 MPa). Although these experiments were operat-
 169 ing in the relatively low-stress range, the apparatus has
 170 the capacity to deliver an axial load of 5 MN, a torque
 171 of 5.6 kNm and a fluid cell pressure of 138 MPa.

172 This lack of data on concrete under high levels of
 173 multiaxial compression at elevated temperature led to
 174 the development of mac^{2T} , the new experimental rig at
 175 The University of Sheffield (Fig. 1).

176 **Test Facility for Multi-Axial Compression of Concrete** 177 **at Elevated Temperature (mac^{2T})**

178 The apparatus for Multi-Axial Compression of Con-
 179 crete at Elevated Temperature (mac^{2T} ; pro-



Fig. 1. mac^{2T} apparatus for multi-axial compression of concrete at elevated temperature

180 nounced *masset*) was designed to satisfy four key testing
181 criteria:

- 182 (i) Multi-axial compression of 100 mm cubic speci-
183 mens up to 400 MPa at any Lode angle
184 ($\sigma_1 \neq \sigma_2 \neq \sigma_3$). Delivery of stresses up to 400
185 MPa was required to allow structural concretes
186 of the type used in existing nuclear power plant
187 reactor and containment vessels (typical uniaxial
188 compressive strength $f_c = 50\text{--}60$ MPa) to be
189 loaded to peak under high levels of hydrostatic
190 confinement (over 100 MPa). The specimen size
191 was fixed at 100 mm to ensure that a represen-
192 tative volume was tested when using 20 mm
193 coarse aggregate. These constraints led to a
194 design solution incorporating three independent
195 loading frames each of 4 MN load capacity.
- 196 (ii) Ability to test in the post-peak range. Valuable
197 information on the effective ductility and frac-
198 ture energy can be obtained by monitoring the
199 post-peak response. However, once the maxi-
200 mum stress is attained, there can be a sudden
201 release of energy stored in the loading frames
202 and the specimen. This can lead to an uncontrol-
203 lable disintegration of the specimen, with loss of
204 load and displacement measurements. In mac^{2T}
205 this effect was minimised by using a combination
206 of compact, stiff loading frames (reducing the
207 elongation of the tensile bars and bending of the

crossheads) and a fast, displacement-controlled 208
servo-hydraulic system that is able to unload the 209
actuators rapidly as the specimen fragments. A 210
disadvantage of rigs where displacements are 211
measured using strain gages bonded to the spec- 212
imen surface (typical for triaxial cells) is that the 213
large cracks that develop in the post-peak range 214
can fracture the gauges, or local spalling occurs 215
leaving the gauge intact but useless in terms of 216
capturing the global response. This is avoided in 217
 mac^{2T} by using un-interruptible laser interferom- 218
eters that allow an accurate, continuous displace- 219
ment signal throughout the test. 220

- (iii) Multi-axial compression at temperatures up to 221
300°C. The high temperature practically elimi- 222
nates the fluid platen option and introduces addi- 223
tional requirements on the data acquisition 224
system, such as minimising the effects of temper- 225
ature variations on strain measurement. In mac^{2T} 226
the loading platens are made of temperature resis- 227
tant steel (Durehete 1055, 20CrMoVTiB4-10), 228
whereas the thermal effects on the strain mea- 229
surements are minimised by using a contactless, 230
laser interferometer system. 231
- (iv) Complex multi-stage experiments following arbi- 232
trary pre-programmed loading paths with simul- 233
taneous temperature cycling. This requirement 234
reflects the desire to generate experimental data 235
for calibrating generalised 3D models able to sim- 236
ulate the response of the material to any combi- 237
nation of loading paths and temperature histories. 238
This was achieved by a custom built system for 239
data acquisition and control with dedicated con- 240
trol software specially developed to meet this 241
requirement. 242

Other important design requirements were: (v) to 243
minimise the friction on the platen-specimen inter- 244
face, and (vi) to ensure that the three stresses are delivered 245
centrally on the six faces of the specimen. 246

Loading and Load Measurement 247

The load in mac^{2T} is delivered by three 4 MN hydraulic 248
actuators installed in independent, diagonally inter- 249
laced loading frames (Fig. 2). This design was chosen to 250
minimise the snap-back potential of the loading frames 251
without increasing the demand on the unloading speed 252
of the servo-hydraulic control system. The high stiff- 253
ness of the frames was achieved by keeping the tensile 254
bars and the crossheads as short as possible, while 255
allowing sufficient room for the specimen to be installed 256
and removed. Each frame comprises two 200 mm 257

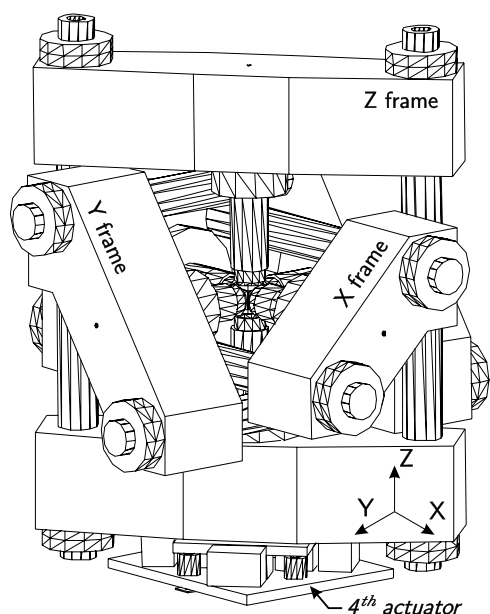


Fig. 2. mac^{2T} loading frames and 4th actuator

258 diameter steel tensile bars and two 550 mm thick, rolled
 259 steel crossheads, one solid, used as a reaction block
 260 (supporting the load cell), the other containing the fluid
 261 chamber for the hydraulic actuator (Fig. 3). This rep-
 262 represents an alternative design solution to that adopted in
 263 the six-actuator 250 kN ASTREE apparatus at LMT-
 264 Cachan [22].

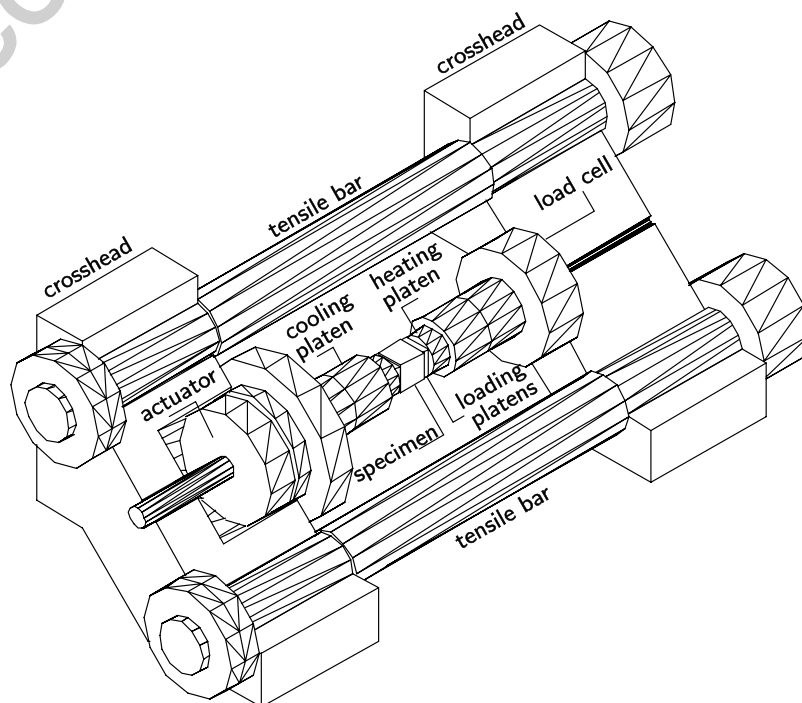
265 The load to the specimen is delivered by 200 mm
 266 diameter rams, through a system of cooling, heating
 267 and loading platens. In each test, a set of six new 1 mm

268 thick steel tiles with 0.25 mm PTFE pads are placed
 269 between the loading platens and the concrete speci-
 270 men to reduce friction-induced shear stresses at the
 271 interface. The loading platens have a spherical seat
 272 arrangement to accommodate minor departures in the
 273 specimen from a right-regular cube. The 95 mm square
 274 loading surface is smaller than the surface of the
 275 specimen to prevent contact between adjacent loading
 276 platens.

277 One of the actuators (X) has a 180 mm stroke which,
 278 when fully retracted, leaves enough space for installing
 279 and removing the specimen. The stroke of the other two
 280 actuators is 60 mm. The pressure in the system is
 281 provided by a 30 MPa hydraulic power unit, which can
 282 generate loads up to 3.61 MN (or $\sigma = 400$ MPa over a
 283 $95 \times 95 \text{ mm}^2$ area).

284 A nominally uniform stress field in a (homogenous)
 285 specimen can only be achieved by (i) eliminating the
 286 friction on the platen-specimen interface, (ii) ensuring
 287 that the central axes of the loading platens always
 288 cross at the centroid of the specimen and (iii) using
 289 precisely machined, right regular cubic specimens. In
 290 previous rigid-platen rigs the friction has been reduced
 291 by using either greased polyethylene pads (New Mexico
 292 [15]) or brush platens (Eindhoven [17]). Polyethylene
 293 pads cannot sustain the high temperature levels
 294 required in mac^{2T} and, although brush platens have
 295 been successfully applied in high-temperature testing
 296 (in the biaxial Braunschweig rig [19]), they would not
 297 be suitable for post-peak testing where large post-
 298 peak deformations may lead to bending and buckling

Fig. 3. Cross-section of loading frame X



299 of the brushes. PTFE remains stable at temperatures
 300 up to 400°C, with very little change in the friction coef-
 301 ficient for temperatures between 20°C and 327°C. The
 302 0.25 mm PTFE pads, eventually selected for mac^{2T},
 303 showed no signs of damage at temperatures up to 300°C
 304 and stresses of up to 330 MPa.

305 It would be inappropriate to attempt a highly de-
 306 tailed FE prediction of the actual stress field operat-
 307 ing within a specimen, as the heterogeneity result-
 308 ing from the stiff aggregate particles (of unknown posi-
 309 tion) creates local stress gradients that mask any non-
 310 uniform effects from the platen-specimen contact. The
 311 decision taken here (common to all such multiaxial ex-
 312 perimental work with structural concrete) has been to
 313 report nominal stress and strain measures when pre-
 314 senting the results.

315 Eccentric loading is prevented by supporting the
 316 two horizontal frames (X and Y) on low friction roller
 317 bearings, which allow free movement of the crossheads
 318 in the horizontal plane. Movement of the centroid of
 319 the specimen in the vertical direction (as a result of
 320 elastic deformations of the load cell, platens and
 321 loading frame) is prevented by active control of the
 322 vertical loading frame (Z) position during the tests.
 323 This is achieved by an array of 4 linked servo-hydraulic
 324 actuators (collectively known as the 4th actuator,
 325 installed beneath the Z frame (see Fig. 2) and con-
 326 trolled by the difference in the displacements mea-
 327 sured at the opposite faces of the specimen. This
 328 system contrasts with the ASTREE rig at LMT-
 329 Cachan [22], where an actuator and a load cell are
 330 used to control/monitor the position/load on each side
 331 of the specimen. mac^{2T} lifts/lowers the entire Z frame
 332 to ensure that the specimen centroid remains at the
 333 same position within the laboratory throughout the
 334 test. Differences in the axial forces from one side of
 335 the specimen to the other are minimised through the
 336 use of the low friction membranes at the platen-
 337 specimen interfaces.

338 The crossheads of the X and Y load frames are
 339 supported on 4 reinforced concrete seats bolted to a
 340 heavily reinforced concrete slab (Fig. 4). The effects of
 341 vibrations induced by machinery operating in the
 342 laboratory are reduced by supporting the slab on three
 343 columns founded at the lower level (basement) and
 344 completely isolated from the floor. Two reinforced
 345 concrete columns and an upper crossbeam support
 346 vertical guides which keep the Z frame in a vertical
 347 plane during re-positioning by the 4th actuator.

348 The applied forces are measured by using three 4
 349 MN rated, high precision (± 4 kN) load cells, installed
 350 on the reaction side of each frame. These are low-
 351 profile (150 mm), shear-type, 400 mm diameter units,

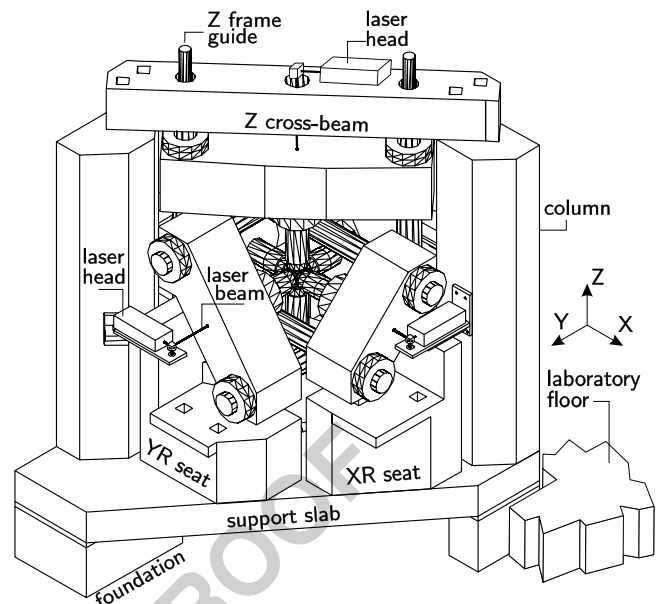


Fig. 4. mac^{2T} load frames and supports

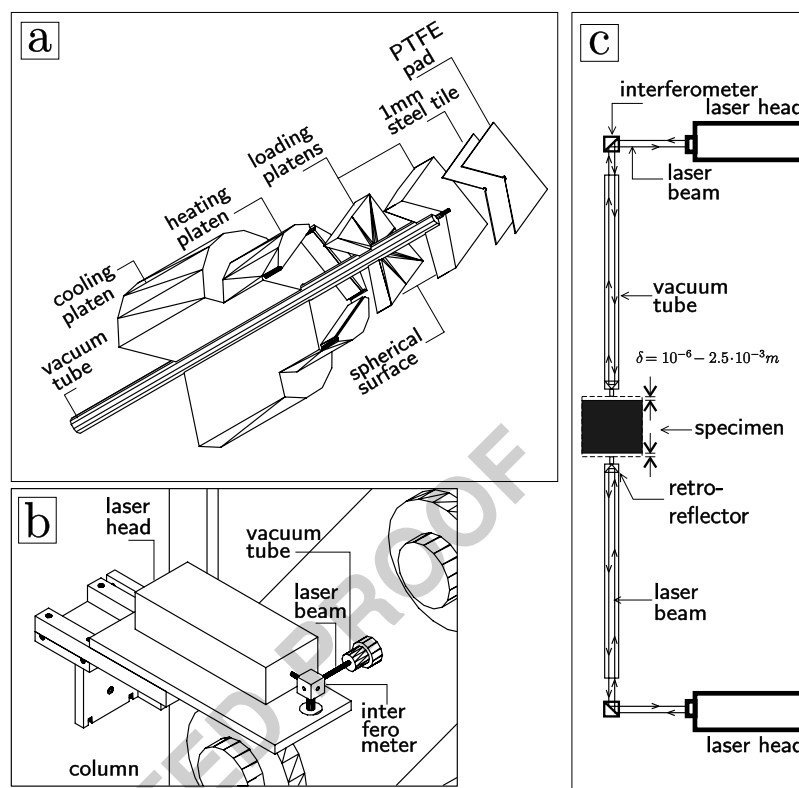
each equipped with two full-bridge strain gauge 352
 arrangements. The signal from each bridge is acquired 353
 independently and then averaged to provide the 354
 process variable for the closed-loop control system 355
 (when an axis is under load control). The three load 356
 cells are calibrated in the rig, using a separate 3.5 MN 357
 load cell, which is itself regularly calibrated at the UK 358
 National Physical Laboratory. 359

Displacement and Strain Measurement 360

Positioning of the actuators, all the strain measure- 361
 ments and the strain control during the tests are 362
 realised by using a system of 6 laser interferometer 363
 units. Each unit comprises a laser head (Hewlett 364
 Packard, model 5529), a linear interferometer, a linear 365
 retro-reflector and a programmable PC calibrator 366
 board installed in a dedicated (slave) PC. 367

368 The retro-reflectors are installed in 18 mm diameter 368
 stainless steel tubes, evacuated to 10^{-5} bar, inserted in 369
 the 25 mm diameter holes which run through the 370
 actuator rams and platens (Fig. 5), as well as through 371
 the load cells and crossheads on the reaction side. The 372
 vacuum is necessary to reduce turbulent movement of 373
 hot air which disturbs the laser beams and may 374
 interrupt the measurements. At the specimen end of 375
 each of these tubes there is a 5 mm diameter steel pin, 376
 which protrudes through the loading platens (including 377
 the 1 mm protective steel tile and 0.25 mm PTFE pad) 378
 and contacts the face of the specimen directly. Within 379
 the tube, the retro-reflector is secured in position at a 380
 distance of 47 mm from the tip of the pin. It is just over 381

Fig. 5 System for strain measurement and displacement control: (a) platens with installed vacuum tubes, (b) mounting of lasers and (c) schematic diagram of laser interferometer system



382 this length that thermal expansions and contractions of
 383 the pin and the tube have to be calibrated-out during
 384 the heating/cooling phases of a test. This setup gives
 385 more accurate and reliable results than systems in
 386 which the strains are calculated from displacements
 387 measured at the ends of long rods [19, 20].

388 This system is capable of measuring large displace-
 389 ments with 10^{-6} m accuracy, making it suitable for
 390 positioning of the actuators (controlling up to 150 mm
 391 of pre-contact movement) and the vertical frame (using
 392 the 4th actuator) as well as for strain measurements and
 393 strain control during the tests (with displacement rates
 394 as low as $10^{-5} \text{ m} \cdot \text{min}^{-1}$). In both cases the values ob-
 395 tained from the displacement measurement system are
 396 used to calculate the process variables for the servo-
 397 control system. Strains in the specimen are calculated
 398 from the difference between the displacements mea-
 399 sured on the opposite faces of the specimen.

400 Heating and Cooling

401 Heat in the rig is generated by a set of six 240 W
 402 ceramic band heaters wrapped around the 140 mm
 403 diameter steel heating platens [Fig. 5(a)]. Heat is
 404 transferred by conduction through the Durehete 1055
 405 (20CrMoVTiB4-10) steel loading platens to the speci-

mens. The load-cell and actuator are kept cool by 406
 flushing cold water through a network of holes cross- 407
 drilled through the 200 mm diameter steel cooling 408
 platens [Fig. 5(a)]. 409

Data Acquisition and Control 410

The system for data acquisition and control of mac^{2T} 411
 is integrated into a PC-based solution built around 412
 National Instruments hardware and MCC (mac^{2T} 413
 Control Centre), a dedicated LabVIEW program for 414
 data acquisition and control developed by the first 415
 author. The master PC used in the tests is a dual Athlon 416
 2 GHz platform equipped with one 24 bit digital input/ 417
 output board (DIO), two 16 bit multifunction boards 418
 (MIO) and one 12 bit analogue output module (AO). 419

All tests in mac^{2T} are monitored by using a total 420
 of 49 sensors: (i) 3 load cells (each with 2 independ- 421
 ently monitored full-bridge sensor arrangements), 422
 (ii) 6 laser interferometer systems for displacement 423
 measurement (both position and specimen deformation) 424
 (iii) 1 pressure transducer for monitoring the 425
 pressure in the 4th actuator, (iv) 6 linear variable 426
 displacement transducers (LVDTs) used as a backup 427
 displacement measurement system, (v) 6 LVDTs 428
 used to monitor the position of the crossheads and 429

430 (vi) 24 K-type thermocouples installed on the heating
431 platens (4 per platen).

432 The signals from all analogue devices (LVDTs, load
433 cell bridges, pressure transducer and thermocouples)
434 are first conditioned, then passed to the analogue IO
435 boards where they are converted into digital data at a
436 rate of $500 \text{ samples}\cdot\text{s}^{-1}$ (multiplexed over 43 channels).
437 In MCC, the acquired data are then averaged in
438 packages of 100 samples at a time. The signals from
439 the laser interferometers are generated as digital data
440 by the six PC calibrator boards installed in the slave
441 PC and transferred to the master PC by the means of
442 digital IO communication controlled from MCC.

443 The averaged input values and the test control param-
444 eters are used for calculating the control variables in the
445 PID (proportional, integral, derivative) modules in
446 MCC. The program outputs are the values of the
447 signals to be sent to the 4 actuator servo-valves and
448 6 temperature controllers. The analogue (voltage)
449 signals are generated by four 16-bit D/A converters of
450 the MIO boards, for the actuators, and six 12-bit D/A
451 converters of the AO module, for the control of the
452 heaters.

453 Operation of the Rig

454 The process of acquiring the data, averaging the
455 values, calculating the control variable and sending
456 the control signal to the servo-valve or temperature
457 controller represents a fully closed control loop. The
458 mac^{2T} rig is operated by 10 independent control loops:
459 3 for the main actuators, 1 for the 4th actuator and 6
460 for the temperature controllers. The actual time for clos-
461 ing the loops depends on the ratio between the sam-
462 pling rate and the number of averages (0.1 s, for the
463 current hardware and software configuration).

464 In essence, the operation of the rig is controlled by
465 defining three sets of parameters for each of the four
466 actuators: (i) control mode, (ii) process variable (PV),
467 and (iii) rate of change of the PV; and the rate of
468 change of the PV for the temperature controllers.

469 Control Loops

470 The three load frame actuators can be operated in
471 either load (LC) or displacement control (DC) mode.
472 The process variable for load control is the load L
473 calculated as an average of the readings from the two
474 bridges of each load cell. The displacement control can
475 be performed by using one of the three different values
476 as process variables: (i) V -the laser measurements of
477 the actuator position, (ii) dV -the laser measurements

of the specimen deformation (difference between the
displacements measured at the two opposite faces of
the specimen), and (iii) U -the LVDT measurements of
the position of the actuators. The diagram of the
control loops is shown in Fig. 6.

The V -control is used for positioning the actuators
before the beginning of the tests (loading) and after the
completion of the tests (unloading). The dV -control is
used for displacement control during the tests. The
 U -control is used in a backup (safety) procedure that is
automatically triggered in case any of the laser signals is
interrupted. Each load frame is controlled indepen-
dently, which means that each of the actuators can be
under L , V , dV or U -control, regardless of the control
mode of the others.

The 4th actuator only operates in displacement con-
trol, using one of the three process variables. The six
heaters are controlled independently with process
variables obtained as average values of the temperature
measurements obtained from the 4 thermocouples
installed on each of the 6 heating platens.

Modes of Operation and Test Stages

The MCC software is designed to provide two basic
modes of operation of the rig: (i) interactive and (ii)
automatic. In the interactive mode, the operator con-
trols the rig either by manually setting the control pa-
rameters (control modes, PV and rates) for each
actuator and temperature controller, or by activating a
range of different pre-programmed procedures. In the
automatic mode, the program reads an input file at the
beginning of the test and performs a series of opera-
tions, without the need for further intervention from
the operator. The automatic procedure can always be
overridden by the operator and switched into interac-
tive mode.

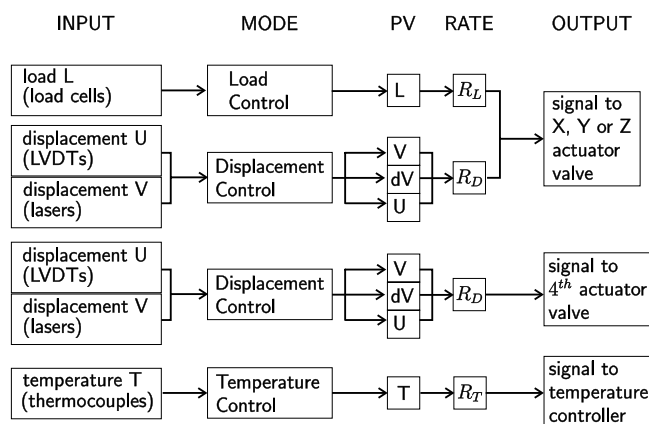


Fig. 6. Operation of mac^{2T} : diagram of control loops

513 The changes between the control modes are *bump-*
 514 *less* and can be performed at any time during the
 515 execution of the test as prescribed by the input file,
 516 triggered by the MCC automatically (in case of certain,
 517 pre-programmed conditions) or by the operator (in
 518 interactive mode). The process variables and rates for
 519 each control loop can be also prescribed in the input
 520 file, defined by the operator during the test or triggered
 521 by the MCC automatically.

522 Each test is performed in three stages: (i) loading,
 523 (ii) testing, and (iii) unloading. Once the specimen is
 524 installed in the rig, the operator activates the auto-
 525 matic loading procedure in which the Z frame is lifted
 526 into position and the three actuators are extended until
 527 the specimen is loaded to contact stress (pre-defined,
 528 typically 0.5–1.0 MPa) in all three directions, first Z,
 529 then X and finally Y. During this stage, the movement
 530 of the Z-frame (4th actuator) and the three loading
 531 actuators is controlled by displacements (V-control)
 532 with rates decreasing gradually as the platens approach
 533 the specimen. Once the contact stress is detected in
 534 any of the three directions, the control of that actuator
 535 switches to load and the rate of the process variable is
 536 set to zero. When all axes are in load control the
 537 operator can start the testing stage. Similarly, when the
 538 test is finished, the unloading procedure starts by
 539 switching the controls to displacement (V) and retract-
 540 ing the actuators in a reverse order: Y, X, Z and finally
 541 the 4th actuator, until the Z-frame rests on its base.

542 In the testing stage the three main actuators are
 543 either under load or dV-control, with load/displace-
 544 ment rates prescribed by the pre-programmed test
 545 procedure or set by the operator. The 4th actuator is
 546 always in dV-control, with variable displacement rates
 547 calculated by an MCC module programmed to keep
 548 the centroid of the specimen at the same level with the
 549 horizontal loading axes during the tests.

550 Programmed Test Sequences

551 All mac^{2T} tests can be composed by using six test
 552 sequences (procedures) programmed in MCC. Each se-
 553 quence contains predefined criteria. When these criteria
 554 are met, the sequence is completed and the program
 555 moves on to the next sequence. The modular structure
 556 of the program allows other sequences to be added in
 557 the future.

558 The load cycling in each test is performed by using
 559 one of the three different *loading/unloading* proce-
 560 dures: 3LC, 1DC or 2DC.

561 1. The 3LC procedure, in which the three main
 562 actuators are controlled independently in load

control, is used for loading/unloading under spec- 563
 564 ified stress conditions at a safe distance from the
 565 peak nominal stress (PNS) surface. The sequence
 566 is completed when the load in one axis (*leader*)
 567 reaches a pre-defined level.

2. 1DC is a procedure in which one axis (σ_1) is under 568
 569 displacement control with a prescribed displace-
 570 ment rate (R_D), whereas the other two axes are in
 571 load control, following σ_1 at preset ratios α_2 and α_3 .
 572 The change of σ_1 is monitored at regular time inter-
 573 vals (t_c), and, assuming that the stress rate re-
 574 mains unchanged during two successive intervals
 575 ($\Delta\sigma_1^{i+1} = \Delta\sigma_1^i$), the loading rates for the other two
 576 directions ($k = 2, 3$) are calculated as: $R_k =$
 577 $[\alpha_k(\sigma_1^i + \Delta\sigma_1^i) - \sigma_k^i]/(t_c A)$, where A is the loading
 578 area. This procedure is used for experiments in
 579 which one stress (σ_1) is close to the peak, such as
 580 triaxial compression ($\sigma_1 > \sigma_2 = \sigma_3$), or tests in which
 581 the three stresses are different ($\sigma_1 > \sigma_2 > \sigma_3$).

3. 2DC is similar to the 1DC procedure, but with two 582
 583 *leader* axes in displacement control (with pre-
 584 scribed rates R_{D1} and R_{D2}) and one in load con-
 585 trol. The loading rate in the third axis is
 586 calculated in the same way as that in the 1DC
 587 procedure, but with the stress rate taken as an
 588 average of the stress increments measured in the
 589 two *leader* axes.

This procedure is used for close-to-peak testing 590
 591 near the extension meridian, where the stresses in
 592 two directions are similar while the stress in the
 593 third direction decreases or remains constant
 594 ($\sigma_1 \approx \sigma_2 > \sigma_3$).

The two DC procedures are used for testing 595
 596 close to peak and in the post-peak region. They are
 597 not suitable for lower stress levels where the
 598 specimen is still relatively stiff and even low
 599 displacement rates in the *leader* axes can produce
 600 high loading rates in the load-controlled axes,
 601 possibly resulting in serious over/under shooting
 602 of the PID controllers and, ultimately, to loss of
 603 control. The execution of the DC sequences is
 604 controlled by setting limits to the values of (i) the
 605 loading stiffness $S = \Delta\sigma_1/\Delta\varepsilon_1$ and (ii) ratio be-
 606 tween the current stress and the maximum stress
 607 recorded during the sequence $\beta = \sigma_1/\hat{\sigma}_1$; calculated
 608 for the *leader* axis (or average of the two *leader*
 609 axes in 2DC) at the end of intervals with a pre-
 610 defined duration t_c . The limit of the stress ratio β
 611 is always set between 0 and 1. If $S_{lim} > 0$, then the
 612 sequence is terminated when $S < S_{lim}$, before the
 613 specimen starts softening (pre-peak tests). If
 614 $S_{lim} < 0$, then the sequence is terminated when the
 615 stress σ_1 drops below $\beta_{lim} \hat{\sigma}_1$ (post-peak tests).

616 *Stress probing procedures* are test sequences in
 617 which the specimen is loaded/unloaded by a small
 618 stress (typically $\Delta\sigma^P = 2 - 3\text{MPa}$) in each of the
 619 three directions, while the strains in the other two
 620 directions are kept constant. The results are used
 621 to determine the values of the tangent stiffness and
 622 elastic unloading stiffness matrices of the material
 623 for a given stress state (at the start of the probing
 624 procedure).

625 4. P3LC is a stress probing procedure in which the
 626 three stress probes are performed in load control.
 627 When the stress in the probing direction i increases
 628 by $\Delta\sigma_i^{pj}$, the strain in this direction changes to
 629 $\varepsilon_i + \Delta\varepsilon_i^{pj}$. The strains in the other two directions
 630 are kept constant ($\Delta\varepsilon_j^{pj} \approx \Delta\varepsilon_k^{pj} \approx 0$) while the stresses
 631 change by $\Delta\sigma_j^{pj}$ and $\Delta\sigma_k^{pj}$, as shown in Fig. 7(a).
 632 When the three probes are completed the tangent
 633 stiffness matrix \mathbf{D} is calculated from the stress and
 634 strain increments measured during the loading
 635 parts of the three probes [see Fig. 7(b)]. The elastic
 636 unloading matrix \mathbf{D}_u is calculated in a similar way
 637 from the values recorded in the unloading branches
 638 of the three probes. After completing the three
 639 probes, the stresses are returned to their initial
 640 levels recorded at the start of the probing sequence,
 641 and the program moves on to the next test
 642 sequence.

643 5. P1DC is a probing sequence in which the loading
 644 part of the probe in one direction (σ_1) is performed
 645 under displacement control with a given rate R_D .
 646 This procedure is used in tests where the major
 647 principal stress is greater than the other two stresses
 648 and close to the PNS ($\sigma_1 > \sigma_2 \geq \sigma_3$). If σ_1 is close
 649 to the peak, the loading stiffness of the specimen in
 650 the probing direction can be very low (often
 651 associated with stress relaxation in the other two
 652 directions) and loading to $\sigma_1 + \Delta\sigma_1^{p1}$ can take a long
 653 time. If the specimen is softening, then σ_1 will decrease
 654 under displacement control. To avoid very slow loading
 655 or softening, the time for the loading part of the probe
 656 is limited, resulting in $\Delta\sigma_1^{p1} < \Delta\sigma^P$
 657 or even $\Delta\sigma_1^{p1} < 0$ (stress relaxation).

658 6. P2DC is similar to P1DC, but with two stress probes
 659 performed under displacement control. This procedure
 660 is used in triaxial extension tests where
 661 $\sigma_1 \approx \sigma_2 > \sigma_3$, and the two larger stresses are close
 662 to the peak stress levels.

663
 664 The temperature in the specimen is controlled in all
 665 test sequences by specifying the heating rate R_H (in $^\circ\text{C} \cdot \text{min}^{-1}$)
 666 and limit temperature T_L . When T_L is reached
 667 the program sets R_H to 0.

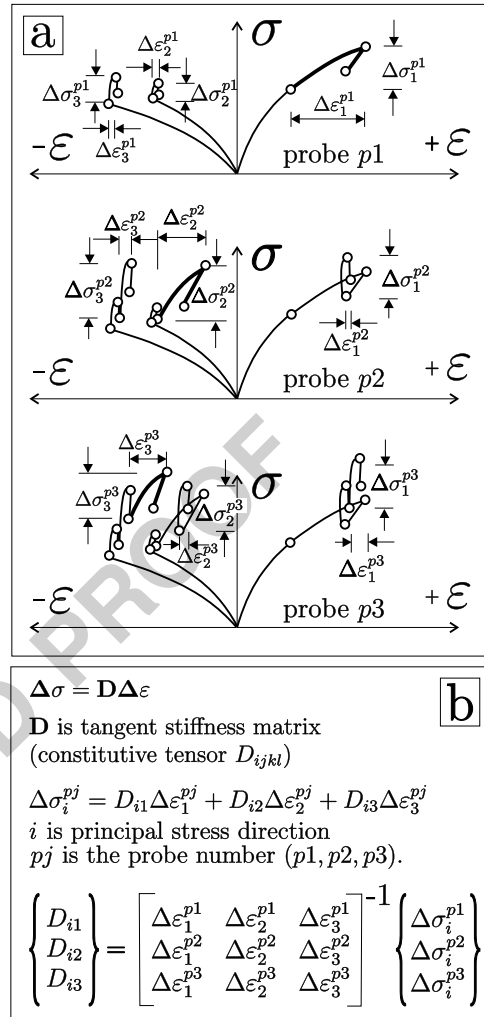


Fig. 7. Stress probing: (a) test sequence and (b) experimental determination of stiffness matrices

Experimental Testing in mac^{2T}

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A typical elevated temperature test in mac^{2T} is performed in two phases: (i) *conditioning* and (ii) *deviatoric loading*.

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Conditioning Phase

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In the *conditioning* phase the specimen is subjected to temperature changes with or without load. When loaded, the specimens are subjected to moderate (service) stresses, typically no higher than 50% of the uniaxial compression strength of the material (f_c). The temperature is either increased monotonically to \hat{T} (up to 300 $^\circ\text{C}$) or applied in cycles between ambient and \hat{T} , at constant heating/cooling rates between 0.1 and 3 $^\circ\text{C} \cdot \text{min}^{-1}$. During the heating and cooling stages the temperature inside the specimen lags behind that measured in the platens. Calibration tests on speci-

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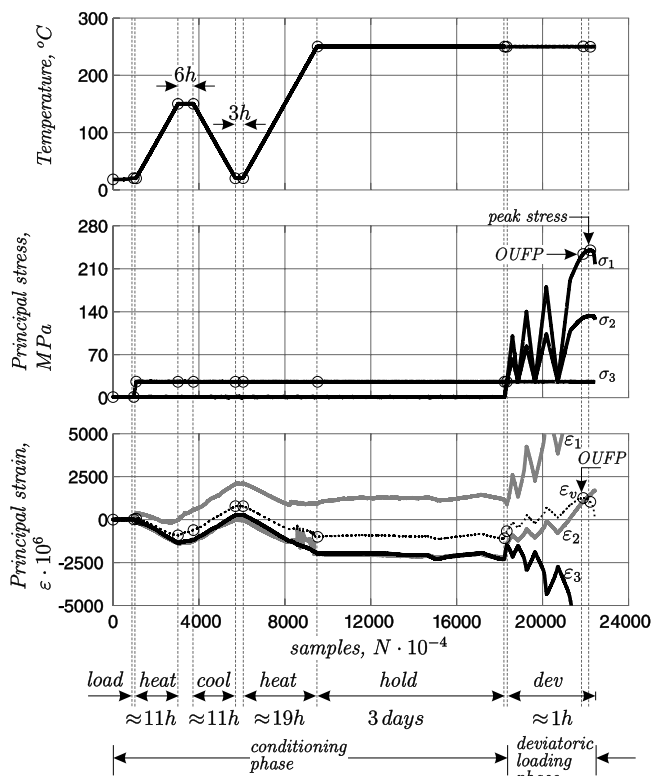


Fig. 8. Typical elevated temperature test in mac^{2T}: histories of temperature, stress and strain

684 mens with embedded thermocouples showed that the
 685 temperatures within the specimen depended on the
 686 heating rate and the moisture content of the concrete.
 687 For 2 year old concrete kept at room conditions,
 688 heated to $\hat{T} = 250^\circ\text{C}$ at $2^\circ\text{C} \cdot \text{min}^{-1}$, the temperatures
 689 within the specimen were 190°C , 10 mm from the
 690 surface, and 180°C , at the centroid, when the platens
 691 first reached 250°C . After maintaining the platen
 692 temperature at 250°C for 2 hours, the specimen centre
 693 reached 236°C . All *conditioning* phases in the elevated
 694 temperature tests performed in mac^{2T} included a period
 695 of at least 24 hours holding at constant temperature of
 696 250°C , at which time the specimen attained a steady
 697 state $\approx 248^\circ\text{C}$.

698 Load-then-heat (L-H) tests are used for investigat-
 699 ing the load induced thermal strains (LITS), a phe-
 700 nomenon of particular importance for fire engineering
 701 and design of concrete structures in nuclear power
 702 plants. After each heating and cooling cycle the
 703 specimen is held under steady state conditions for
 704 short periods until it reaches thermal and hygral equi-
 705 librium. In some tests the heating to \hat{T} is followed by
 706 longer periods of steady state (up to 5 days), in order
 707 to measure creep at constant (elevated) temperature.
 708 Heat-then-load (H-L) tests are used to determine
 709 drying shrinkage (needed for separating LITS from

total strains) and to provide a reference for assessing
 the effects of heat-load regimes on the material be-
 haviour under multiaxial stress.

The time histories of temperature, stress and strain
 from a typical mac^{2T} L-H test are shown in Fig. 8. The
 specimen is first loaded in uniaxial compression to 26
 MPa, then heated to 150°C , cooled to 20°C , heated to
 250°C , and held under steady-state conditions for
 3 days before starting the *deviatoric loading* phase.

Deviatoric Loading Phase

The influence of different loading and heating-cooling
 regimes in the *conditioning* phase on the properties of
 the material under multiaxial loading are investigated
 in the second, *deviatoric loading*, phase of the tests.

In the simplest tests, the specimen is loaded in one
 deviatoric plane (between Lode angles $\pi/6$ and $-\pi/6$),
 in several cycles with gradually increasing stress

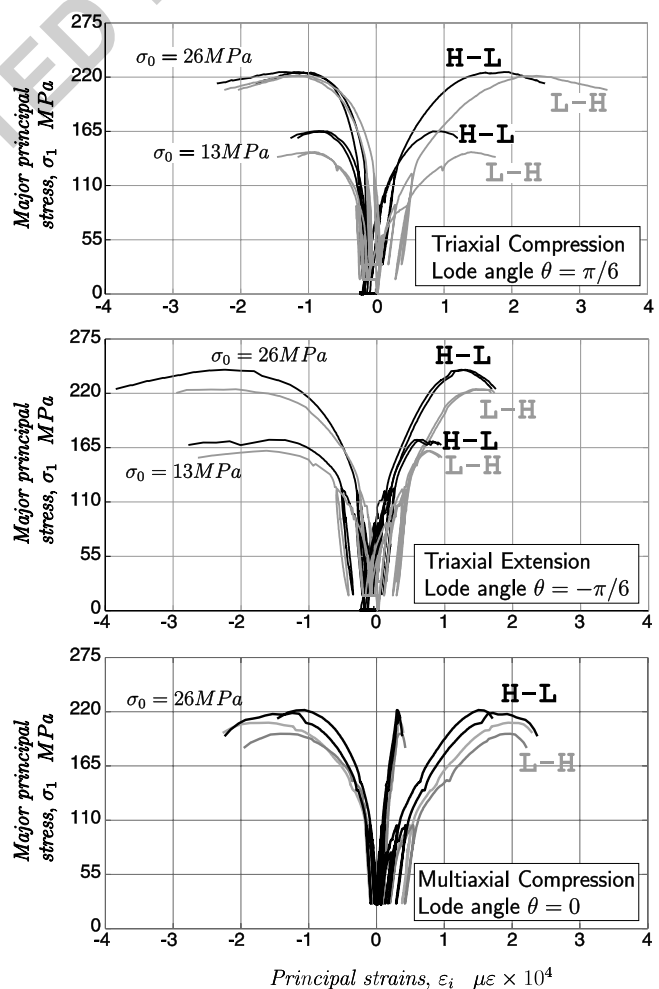


Fig. 9. Simple deviatoric loading test in mac^{2T}: stress-strain graphs recorded after heat-load (H-L) and load-heat (L-H) conditioning at two different levels of hydrostatic confinement σ_0

727 magnitudes, followed by monotonic loading to peak
 728 stress (Fig. 9). These tests provide data on the re-
 729 lationship between heating, cooling and loading (\dot{T} ,
 730 heating-cooling rates, heat-load sequence, and stress
 731 state during heating-cooling) and multiaxial strength of
 732 the material [23], tangent stiffness, elastic unloading
 733 stiffness, pre-peak volumetric expansion (or OUFP—
 734 onset of unstable fracture propagation), plastic flow
 735 and softening behaviour of the material.

736 More complex deviatoric loading tests include (i)
 737 deviatoric load cycles at fixed Lode angles but varying
 738 hydrostatic confinement levels and (ii) stress probes at
 739 different stress levels in each deviatoric cycle.

740 In Fig. 10 are shown the loading path and stress-
 741 strain response recorded in a triaxial compression test
 742 ($\theta = \pi/6$, $\sigma_1 > \sigma_2 = \sigma_3$). The load was applied in six
 743 deviatoric planes (normal to the hydrostatic axis ξ), at
 744 increasing levels of hydrostatic confinement. The test
 745 in each cycle i was carried out in 4 steps: (1) load
 746 hydrostatically to a predefined level σ_0^i , (2) load in a
 747 deviatoric plane to $\sigma_1 = 0.975\hat{\sigma}^i$, (3) perform probes at

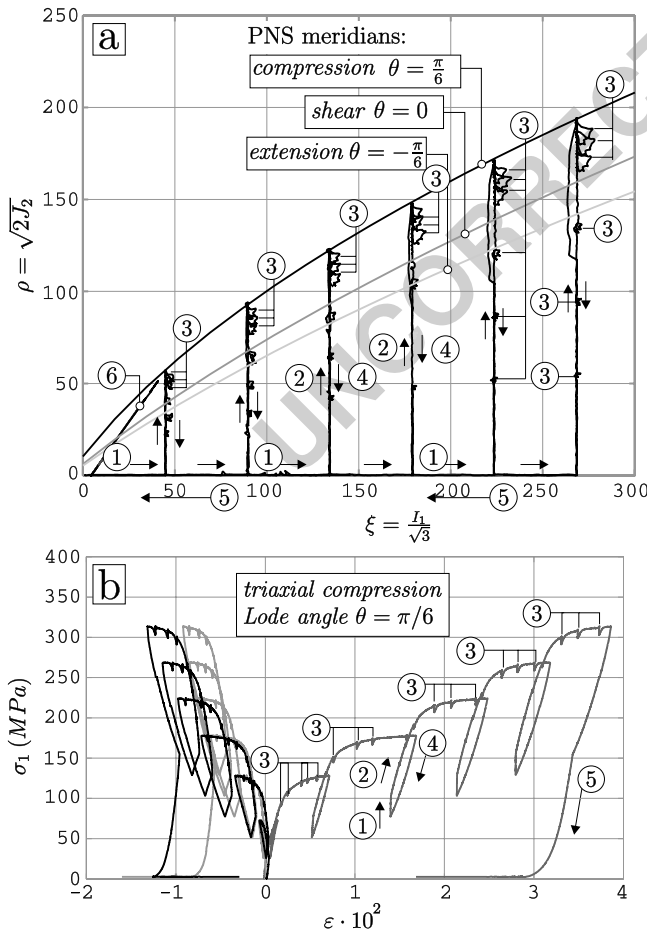


Fig. 10. Complex deviatoric loading in mac^{2T}: (a) loading path and (b) stress-strain response of the specimen under multiaxial compression. $I_1 = \text{tr}[\sigma]$ and $2J_2 = \text{tr}[\sigma^2]$

$$Q_{jkl} = n_i D_{ijkl} n_i$$

Q_{jkl} is acoustic tensor

D_{ijkl} is constitutive tensor

n_i is direction vector

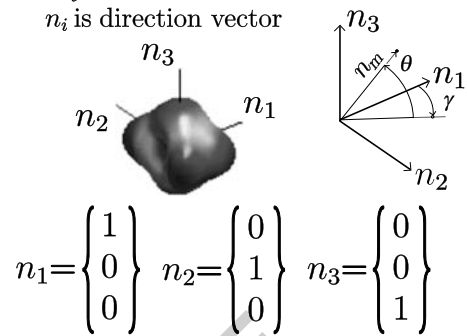


Fig. 11. Acoustic tensor surface calculated from experimentally determined stiffness matrices

6 stress levels $\sigma_1 = 0.6, 0.7, 0.8, 0.9, 0.925$ and $0.95\hat{\sigma}^i$,
 and (4) unload to σ_0^i . After completing the six loading
 cycles, the specimen was unloaded along the hydro-
 static axis (step 5) and tested under σ_3 compression
 ($\sigma_1 = \sigma_2 = 3\text{MPa}$) to determine the residual strength
 of the material (step 6).

Performing a series of small compression *probes* in
 each of the three orthogonal directions, while main-
 taining strains fixed in the other directions, allows the
 isotropic upper 3×3 sub-matrix of the 6×6 constitu-
 tive [D] matrix to be estimated [Fig. 7(b)]. The re-
 maining terms of the [D] matrix were calculated by
 adopting the original elastic shear modulus. This could
 have been estimated, for an equivalent isotropic ma-
 terial, from the altered mean Young's modulus and
 mean Poisson's ratio. However, the effect of not
 changing the shear modulus was shown to be near-
 negligible when examining an equivalent elasto-plas-
 ticity model. By calculating the acoustic tensor from
 this matrix, a measure of impending material instabil-
 ity can be determined [24]. This measure (the deter-
 minant of the acoustic tensor) has both direction and
 magnitude, thus it may be represented graphically
 (Fig. 11) by a surface which starts out as being spheri-
 cal and evolves (collapses) into a form where the radius
 becomes zero in the direction normal to the orientation
 of a newly formed discontinuity surface. Such results
 give a detailed picture of the material behaviour under
 multiaxial compression needed for construction of
 advanced constitutive models.

Conclusions

The new mac^{2T} facility was developed to overcome the
 lack of high quality experimental data on the behav-

781 iour of concrete under true triaxial compression,
 782 under relatively high levels of confinement, at both
 783 ambient and elevated temperature. The fully automat-
 784 ed programmable control system allows multi-stage
 785 experiments to be carried out by following complex
 786 load paths and temperature histories, at any Lode
 787 angle and in the post-peak range. These tests provide
 788 new macroscopic data needed to construct advanced
 789 constitutive models for simulating the response of
 790 concrete under generalised stress states. The observed
 791 effects of different heating and loading regimes on the
 792 evolving mechanical properties provide insight into the
 793 stress and temperature-induced changes in the materi-
 al fabric at a lower, microscopic, level.

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 799

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